

Space-Time Multigrid Techniques for Solving Time-Dependent Optimal Control Problems in CFD

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Distributed Control of nonstationary flow

Distributed Control for the nonstationary Navier-Stokes equation with tracking-type functional for a given z :

$$J(y, u) = \frac{1}{2} \|y - z\|_Q^2 + \frac{\alpha}{2} \|u\|_Q^2 + \frac{\gamma}{2} \|y(T) - z(T)\|_{\Omega}^2 \rightarrow \min!$$

on $Q = \Omega \times [0, T]$ such that

$$\begin{aligned} y_t - \nu \Delta y + (y \nabla) y + \nabla p &= u && \text{in } Q \\ -\nabla \cdot y &= 0 && \text{in } Q \end{aligned} \quad + \text{BC}$$

No constraints on the control $u \in Q$.

Distributed Control of nonstationary flow

Corresponding KKT-System:

$$\begin{aligned} y_t - \nu \Delta y + (y \nabla) y + \nabla p &= u && \text{in } Q \\ -\nabla \cdot y &= 0 && \text{in } Q \end{aligned}$$

$$\begin{aligned} -\lambda_t - \nu \Delta \lambda - (y \nabla) \lambda + (\nabla y) \lambda + \nabla \xi &= (y - z) && \text{in } Q \\ -\nabla \cdot \lambda &= 0 && \text{in } Q \end{aligned}$$

$$u = -\frac{1}{\alpha} \lambda \quad \text{in } Q$$

- + boundary conditions
- + initial condition
- + terminal condition $\lambda(T) = \gamma(y(T) - z(T))$ in Ω

Distributed Control of nonstationary flow

Corresponding KKT-System:

$$\begin{aligned} y_t - \nu \Delta y + (y \nabla) y + \nabla p &= -\frac{1}{\alpha} \lambda && \text{in } Q \\ -\nabla \cdot y &= 0 && \text{in } Q \end{aligned}$$

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Distributed Control of nonstationary flow

Corresponding KKT-System:

$$\begin{aligned} y_t - \nu \Delta y + (y \nabla) y + \nabla p + \frac{1}{\alpha} \lambda &= 0 && \text{in } Q \\ -\nabla \cdot y &= 0 && \text{in } Q \end{aligned}$$

$$\begin{aligned} -\lambda_t - \nu \Delta \lambda - (y \nabla) \lambda + (\nabla y) \lambda + \nabla \xi - y &= -z && \text{in } Q \\ -\nabla \cdot \lambda &= 0 && \text{in } Q \end{aligned}$$

Distributed Control of nonstationary flow

Corresponding KKT-System:

$$\begin{aligned} y_t + N(y)y + \nabla p + \frac{1}{\alpha}\lambda &= 0 && \text{in } Q \\ -\nabla \cdot y &= 0 && \text{in } Q \end{aligned}$$

$$\begin{aligned} -\lambda_t + N^*(y)\lambda + \nabla \xi - y &= -z && \text{in } Q \\ -\nabla \cdot \lambda &= 0 && \text{in } Q \end{aligned}$$

Distributed Control of nonstationary flow

Corresponding KKT-System:

$$\begin{pmatrix} y_t \\ -\lambda_t \\ 0 \\ 0 \end{pmatrix} + \begin{pmatrix} N(y) & \frac{1}{\alpha} & \nabla & \nabla \\ -I & N^*(y) & & \\ -\nabla \cdot & & 0 & \nabla \\ & -\nabla \cdot & & 0 \end{pmatrix} \begin{pmatrix} y \\ \lambda \\ p \\ \xi \end{pmatrix} = \begin{pmatrix} 0 \\ -z \\ 0 \\ 0 \end{pmatrix}$$

$$\Rightarrow \begin{pmatrix} w_t \\ 0 \end{pmatrix} + \begin{pmatrix} N(w) & \nabla \\ -\nabla \cdot & 0 \end{pmatrix} \begin{pmatrix} w \\ q \end{pmatrix} = \begin{pmatrix} f \\ 0 \end{pmatrix}$$

\Rightarrow generalised Navier–Stokes equation

Space-time discretisation

- Time discretisation:

Backward Euler (later: 2nd order Crank-Nicolson, FS- θ)

$$\frac{y_i - y_{i-1}}{\Delta t} + N(y_i)y_i + \nabla p_i + \frac{1}{\alpha}\lambda_i = \dots$$
$$\frac{\lambda_i - \lambda_{i+1}}{\Delta t} + N^*(y_i)\lambda_i + \nabla \xi_i - y_i = \dots$$

- Space discretisation:

LBB-stable Finite Elements (\tilde{Q}_1/Q_0 , later Q_2/P_1)
on general 2D meshes (later: 3D)

Space-time discretisation

⇒ Space-time system

$$A(x)x = b$$

Here (for n timesteps, $x_k = x(t_k)$):

$$x = (\underbrace{y_0, \lambda_0, p_0, \xi_0}_{x_0=x(t_0)}, \dots, \underbrace{y_n, \lambda_n, p_n, \xi_n}_{x_n=x(t_n)})$$

Space–time discretisation

$A(x)$ has a very special structure! E.g. for 2 timesteps:

$$A(x)x = \left(\begin{array}{cc|cc|c} M & -B & -M & -B & y_0 \\ -M & NST^* & 0 & -B & \lambda_0 \\ \hline -B^T & 0 & -\frac{M}{\Delta t} & & p_0 \\ -B^T & 0 & & & \xi_0 \\ \hline -\frac{M}{\Delta t} & & & & y_1 \\ & NST & \frac{M}{\alpha} & -B & \lambda_1 \\ & -M & NST^* & -B & p_1 \\ \hline & -B^T & 0 & -\frac{M}{\Delta t} & \xi_1 \\ & & -B^T & 0 & y_2 \\ \hline & -\frac{M}{\Delta t} & & & \lambda_2 \\ & & NST & -B & p_2 \\ & & -\frac{\gamma M}{\Delta t} & NST^* & \xi_2 \\ \hline & -B^T & 0 & -B^T & y_3 \\ & & -B^T & 0 & \lambda_3 \\ \hline \end{array} \right)$$

→ Sparse, (block) tridiagonal system

Design of a One-Shot solver

- Nonlinearity: Newton method for quadratic convergence

$$x^{i+1} = x^i + A'^{-1}(x^i)(b - A(x^i)x^i)$$

- Linear subproblems: space-time Multigrid solver

→ Convergence rates independent of refinement level of the space-time mesh

Space-time multigrid ingredients

Essential multigrid components:

- Mesh hierarchy for a space-time cylinder $Q = \Omega \times [0, T]$:
Choose arbitrary space-time coarse mesh and refine!
- Prolongation/Restriction in space + time.
Combination of FE in space & FD in time.
- An efficient smoother! E.g. with $\tilde{A} := A'(x^i)$:

$$v^{j+1} = v^j + \omega P^{-1}(b - \tilde{A}v^j), \quad j = 1, \dots, \text{NSM}$$

What to use as preconditioner P ?

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What to use as preconditioner P ?

Space-time discretisation and preconditioner

\tilde{A} in compressed form (omitting B and B^T here):

$$\left(\begin{array}{c|c|c|c} M & & & \\ -M & NST^* & & -\frac{M}{\Delta t} \\ \hline -\frac{M}{\Delta t} & & NST & \frac{1}{\alpha}M \\ & & -M & NST^* \\ \hline & -\frac{M}{\Delta t} & & NST \\ & & & -M \\ & & & NST^* \\ \hline & & & \dots \\ & & & \dots \end{array} \right)$$

⇒ Block-Jacobi preconditioner:

$$P := P_{Jac} := \left(\begin{array}{c|c|c|c} M & & & \\ -M & NST^* & & \\ \hline & & NST & \frac{1}{\alpha}M \\ & & -M & NST^* \\ \hline & & & NST \\ & & & -M \\ & & & NST^* \\ \hline & & & \dots \\ & & & \dots \end{array} \right)$$

Space-time discretisation and preconditioner

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Space-time discretisation and preconditioner

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\Rightarrow Symmetric Gauß-Seidel-like preconditioner $P := P_{SGS}$.

In every timestep:

$$\left(\begin{array}{cc} -\frac{M}{\Delta t} & 0 \\ 0 & 0 \end{array} \right) v_{i-1} + \left(\begin{array}{cc} \textcolor{red}{NST} & \frac{1}{\alpha} M \\ -M & \textcolor{red}{NST^*} \end{array} \right) v_i + \left(\begin{array}{cc} 0 & 0 \\ 0 & -\frac{M}{\Delta t} \end{array} \right) v_{i+1} = \dots$$

$\rightarrow v_i$

Space-time preconditioner

Essential: To use $P_{Jac/SGS}^{-1}$, apply for every timestep:

$$\left(\begin{array}{cc|cc} NST & \frac{1}{\alpha}M & B & B \\ -M & NST^* & B & B \\ \hline B^T & & 0 & 0 \\ & B^T & 0 & 0 \end{array} \right)^{-1}$$

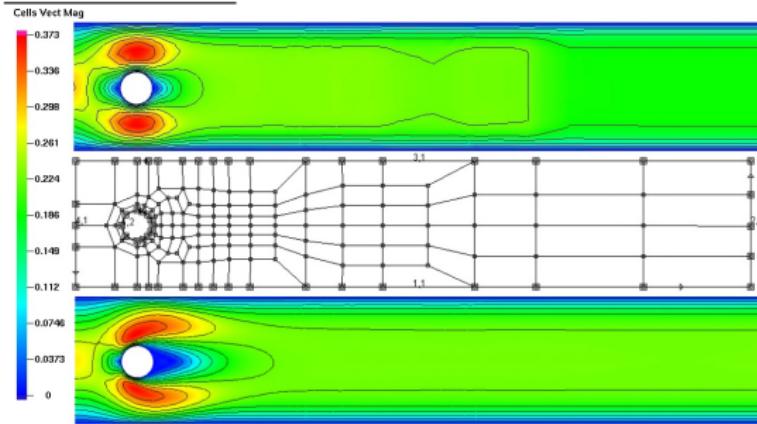
Generalised Navier–Stokes \Rightarrow use efficient CFD techniques!

- Multigrid in space
- Smoother: mod. LPSC (VANKA-like)
→ Primal and dual coupled!

} **FeatFlow!**

Numerical example: Flow-around-cylinder

- Target flow z : Stokes flow, $t \in [0, 1]$, starting from rest



Stokes at $t = 1$

Mesh

uncontrolled Nav.St.

- Optimal control problem: Navier–Stokes, $Re = 20$
- Coarse mesh: 1404 DOF's in space, 5 timesteps, $\Delta t := 0.2$
 $\Rightarrow 8\,424$ DOF's, $\times 8$ per level

Numerical example: Flow-around-cylinder

Convergence of the Newton solver

		Simulation				Optimisation	
Δt	Space-Lv.	#NL	#MG	\oslash #NL	\oslash #MG	#NL	#MG
1/20	3	63	312	3	16	4	24
1/40	4	123	709	3	18	4	14
1/80	5	246	1589	3	20	4	13

- Space-time MG gained 2 digits per step
- Space-preconditioner gained 2 digits per step

Numerical example: Flow-around-cylinder

Convergence of the Newton solver

Δt	Space-Lv.	Time sim.	Time opt.	Time opt / Time sim.
1/20	3	27	917	34.0
1/40	4	209	4346	20.7
1/80	5	2227	55174	24.8

Optimisation is $C \approx 20 - 30 \times$ more expensive than simulation
→ independent of the refinement level

→ Aim: Factor 10-15 due to "code optimisation"!

Numerical example: Flow-around-cylinder

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Numerical example: Flow-around-cylinder

Convergence of the solver for different α and γ

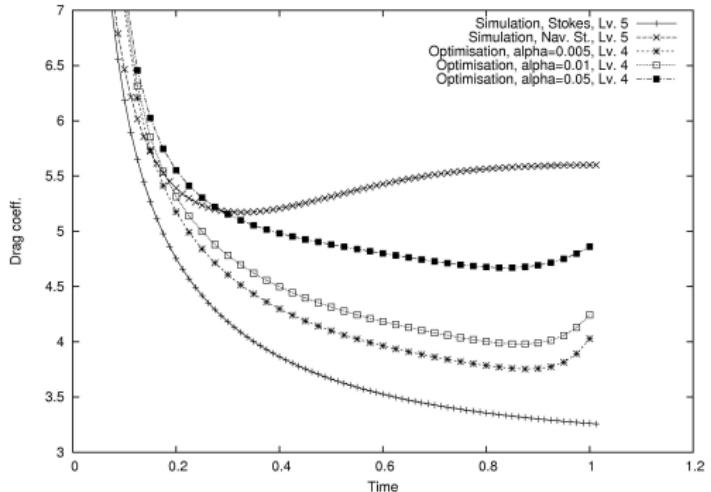
α	γ	#NL	#MG
0.05	0	5	16
	0.1	5	16
	0.5	5	16
0.01	0	5	15
	0.1	5	15
	0.5	5	37
0.005	0	5	22
	0.1	5	16
	0.5	5	72

$\Delta t = 1/40$,
Space-level 4

⇒ Smaller α + larger $\gamma \rightarrow$ worse convergence rates

Numerical example: Flow-around-cylinder

Solution quality: Analysis of the drag

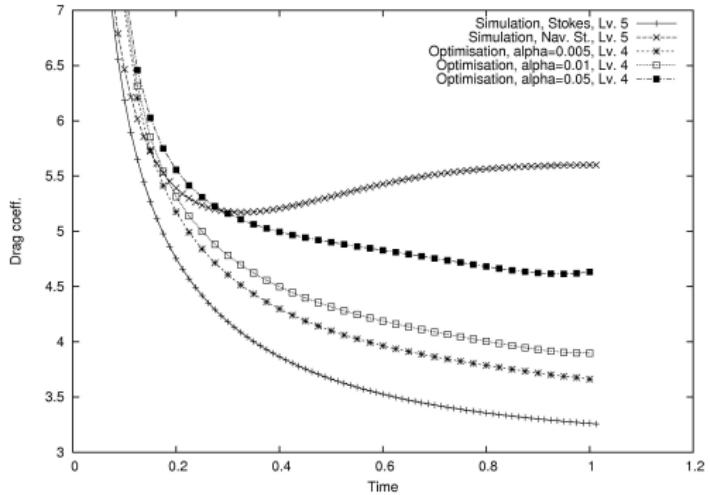


$$\begin{aligned}\alpha &= 0.05, 0.01, 0.005, \\ \gamma &= 0.00\end{aligned}$$

$$C_D = \frac{2}{(\frac{2}{3} U_{\max})^2 \cdot 0.1} \int_{\Gamma} (\sigma n) n_x d\Gamma$$

Numerical example: Flow-around-cylinder

Solution quality: Analysis of the drag

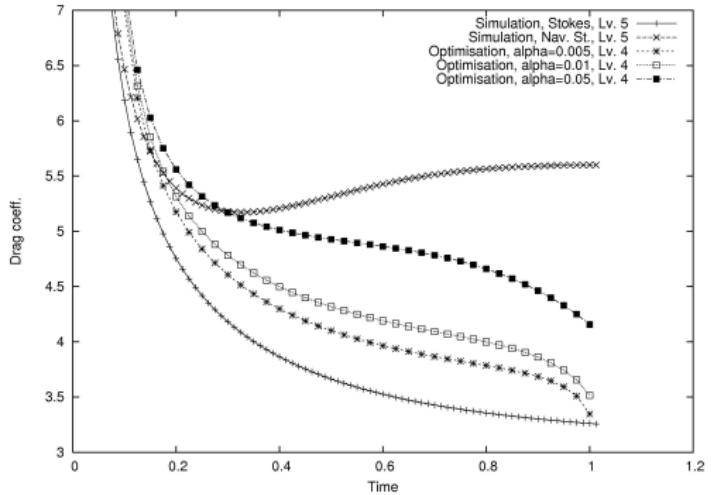


$$\begin{aligned}\alpha &= 0.05, 0.01, 0.005, \\ \gamma &= 0.1\end{aligned}$$

$$C_D = \frac{2}{(\frac{2}{3} U_{\max})^2 \cdot 0.1} \int_{\Gamma} (\sigma n) n_x d\Gamma$$

Numerical example: Flow-around-cylinder

Solution quality: Analysis of the drag



$$\alpha = 0.05, 0.01, 0.005, \\ \gamma = 0.5$$

$$C_D = \frac{2}{(\frac{2}{3} U_{\max})^2 \cdot 0.1} \int_{\Gamma} (\sigma n) n_x d\Gamma$$

Outlook

Further steps:

- Q_2/P_1 + Crank-Nicolson
 - higher accuracy, larger timesteps, fewer unknowns
- FEM-stabilisation (EO-FEM/interior penalty)
 - higher RE numbers
- improved smoothers for the space-time problem
- incorporate constraints
- include algorithms for memory management

Space-time discretisation

Structure of $A(x)$:

$$\begin{array}{c|cc|cc|c}
 M & & -B & & & \\
 -M & \frac{M}{\Delta t} + N^* & -B & & & \\
 \hline
 -B^T & 0 & & -\frac{M}{\Delta t} & & \\
 -B^T & 0 & & & & \\
 \hline
 -\frac{M}{\Delta t} & & & & & \\
 & \frac{M}{\Delta t} + N & \frac{1}{\alpha} \frac{M}{\Delta t} + N^* & -B & & -\frac{M}{\Delta t} \\
 & -M & \frac{\gamma}{\Delta t} M & -B & & \\
 & -B^T & 0 & 0 & & \\
 \hline
 & -\frac{M}{\Delta t} & & & & \\
 & & \frac{M}{\Delta t} + N & \frac{1}{\alpha} \frac{M}{\Delta t} + N^* & -B & -B \\
 & & -\frac{\gamma}{\Delta t} M & \frac{\gamma}{\Delta t} M & 0 & -B \\
 & & -B^T & -B^T & 0 & 0
 \end{array}$$

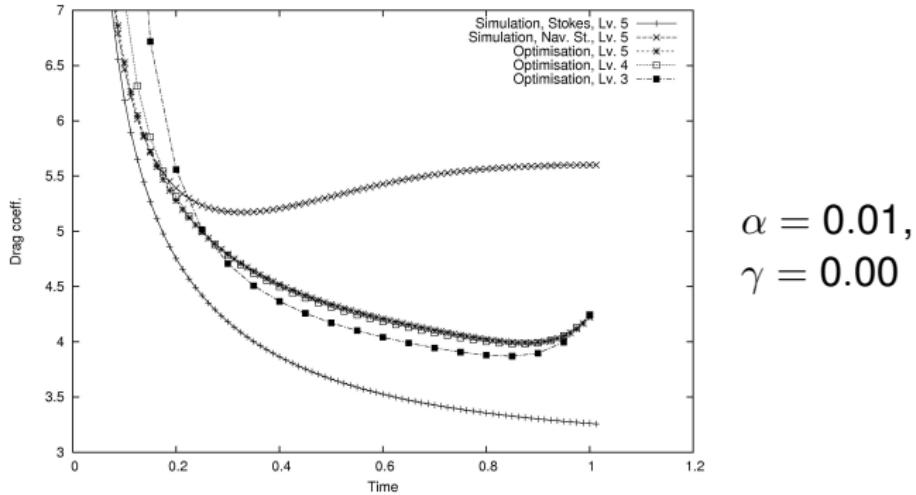
Space-time discretisation

Structure of $A'(x)$:

M	$-B$	$-B$	$-\frac{M}{\Delta t}$	
$-M + R_0$	$\frac{M}{\Delta t} + N^*$	0	$-\frac{M}{\Delta t}$	
$-B^T$	$-B^T$	0		
$-\frac{M}{\Delta t}$	$\frac{M}{\Delta t} + N^{**}$	$\frac{1}{\Delta t} M$	$-B$	$-\frac{M}{\Delta t}$
	$-M + R_1$	$\frac{\dot{M}}{\Delta t} + N^*$	$-B$	
	$-B^T$	0		
	$-\frac{M}{\Delta t}$	$\frac{M}{\Delta t} + N^{**}$	$\frac{1}{\Delta t} M$	$-B$
	$-\frac{\gamma}{\Delta t} M + R_2$	$\frac{\dot{M}}{\Delta t} + N^*$	0	$-B$
	$-B^T$	$-B^T$		0

Numerical example: Flow-around-cylinder

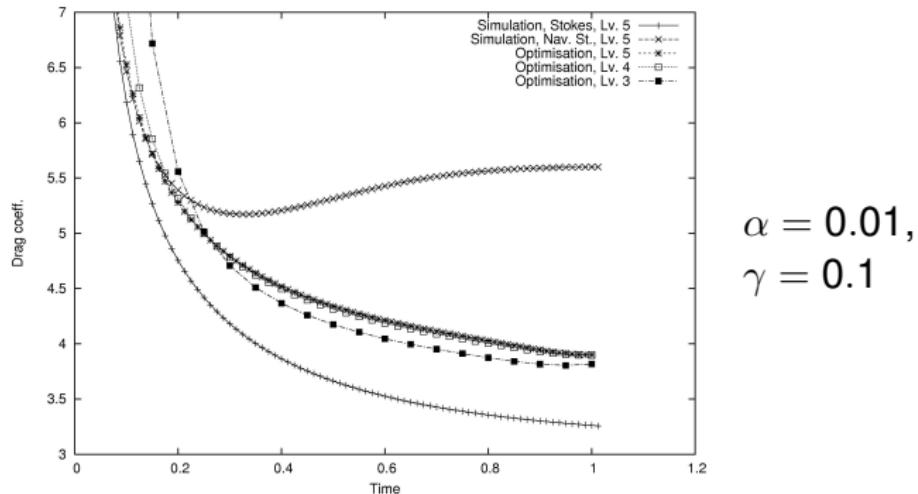
Solution quality: Analysis of the drag



$$C_D = \frac{2}{(\frac{2}{3} U_{\max})^2 \cdot 0.1} \int_{\Gamma} (\sigma n) n_x d\Gamma$$

Numerical example: Flow-around-cylinder

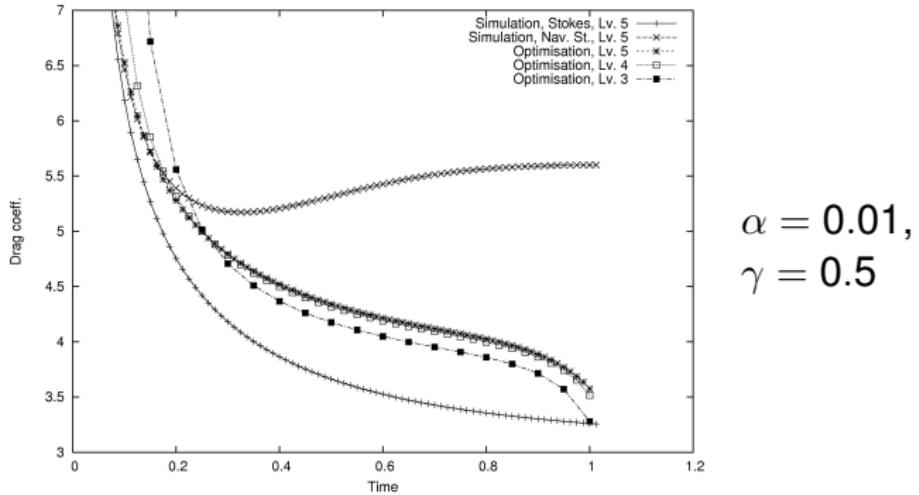
Solution quality: Analysis of the drag



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Numerical example: Flow-around-cylinder

Solution quality: Analysis of the drag



$$C_D = \frac{2}{(\frac{2}{3} U_{\max})^2 \cdot 0.1} \int_{\Gamma} (\sigma n) n_x d\Gamma$$

Numerical example: Flow-around-cylinder

Convergence of the solver for different ν

		$\nu = 1/1000$		$\nu = 1/250$	
Δt	Space-Lv.	#NL	#MG	#NL	#MG
1/10	2	4	36	4	13
1/20	3	4	24	3	8
1/40	4	4	14	3	7
1/80	5	4	13	3	8

⇒ Stabilisation and stronger smoothers necessary
for higher RE numbers

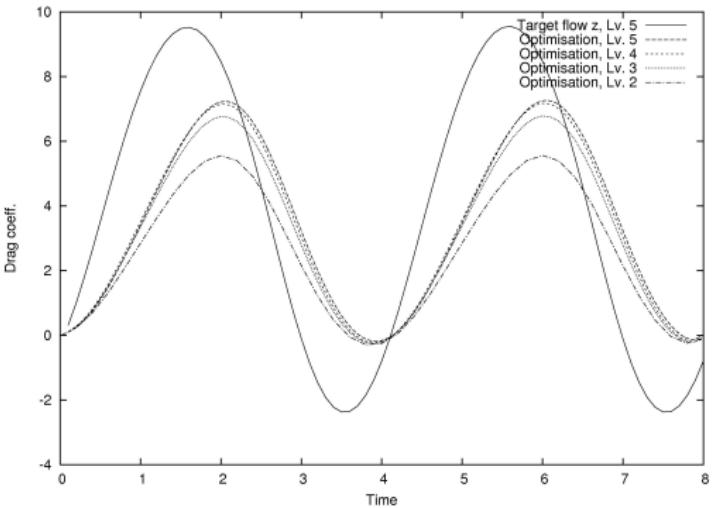
Numerical example:

Flow-around-cylinder with pulsating BC

- Domain: Flow-around-cylinder (like above), $T = 8$.
- Coarse mesh:
1404 DOF's in space, 20 timesteps with $\Delta t = 0.4$.
- Target flow z : Parabolic inflow with max. velocity

$$U_{\max}(t) := 0.15 \cdot \left(1 + \sin\left(\frac{t+3}{2}\pi\right)\right)/2$$

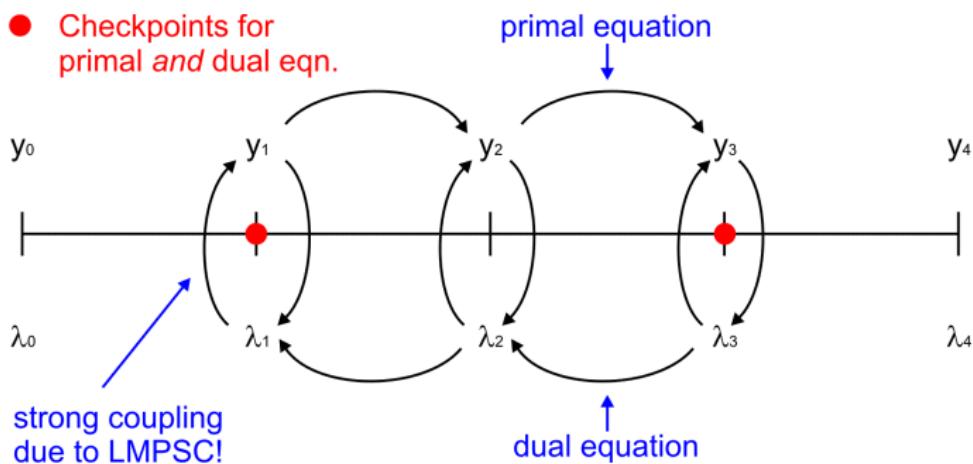
- Optimisation: Natural boundary conditions at inflow



$$C_D = \frac{2}{(\frac{2}{3}0.15)^2 \cdot 0.1} \int_{\Gamma} (\sigma n) n_x d\Gamma$$

Checkpointing in the One-shot approach

- Checkpoints for primal *and* dual eqn.



- Checkpoints → nonlinear subproblems of the same kind.
- High computational costs necessary for recomputation
→ due to strong coupling by LMPSC!